# GRADUATE AERONAUTICAL LABORATORIES CALIFORNIA INSTITUTE OF TECHNOLOGY

## Excimer Laser System for Hydrocarbon-Radical imaging

Paul E. Dimotakis John K. Northrop Professor of Aeronautics And Professor of Applied Physics dimotakis@caltech.edu

Air Force Office of Scientific Research Grant No. F49620-99-1-0328 (Equipment)

DURIP/Instrumentation Grant Final Technical Report for the period 1 September 1999 to 31 December 1999

23 June 2000

20000712 01

MAIL CODE 301-46 PASADENA, CA 91125

### REPORT DOCUMENTATION PAGE

AFRL-SR-BL-TR-00-

Public reporting burden for this collection of information is estimated to average 1 hour per response, including the time maintaining the data needed, and completing and reviewing this collection of information. Send comments regarding this suggestions for reducing this burden to Department of Defense, Washington Headquarters Services, Directorate for Info Suite 1204, Arlington, VA 22202-4302. Respondents should be aware that notwithstanding any other provision of law

0375

ig and ion, including s Highway, th a

collection of information if it does not display a currently	valid OMB control number. PLEASE DO NOT RETURN YOUR FORM TO	THE ABOVE ADDITION.
1. REPORT DATE (DD-MM-YYYY)	2. REPORT TYPE	3. DATES COVERED (From - To)
26-06-2000	Final Technical	1 Sep 1999 - 31 Dec 1999
4. TITLE AND SUBTITLE		5a. CONTRACT NUMBER
(U) Excimer Laser System for		
•		5b. GRANT NUMBER
		F49620-99-1-0328
		5- PROCEAN ELEMENT NUMBER
		5c. PROGRAM ELEMENT NUMBER 61102F
		61102F
6. AUTHOR(S)	5d. PROJECT NUMBER	
		2308
		5e. TASK NUMBER
Paul E. Dimotakis	BX	
Paul E. Dimotakis		5f. WORK UNIT NUMBER
		JI. WORK ONLY NOMBER
7. PERFORMING ORGANIZATION NAME(S	S) AND ADDRESS(ES)	8. PERFORMING ORGANIZATION REPORT
	S) AND ADDRESS(ES)	8. PERFORMING ORGANIZATION REPORT NUMBER
California Institute of	S) AND ADDRESS(ES)	
	S) AND ADDRESS(ES)	
California Institute of	S) AND ADDRESS(ES)	
California Institute of Technology	S) AND ADDRESS(ES) .	
California Institute of Technology	S) AND ADDRESS(ES)	
California Institute of Technology Pasadena CA 91125		NUMBER
California Institute of Technology Pasadena CA 91125  9. SPONSORING / MONITORING AGENCY		
California Institute of Technology Pasadena CA 91125  9. SPONSORING / MONITORING AGENCY AFOSR/NA		NUMBER
California Institute of Technology Pasadena CA 91125  9. SPONSORING / MONITORING AGENCY AFOSR/NA 801 North Randolph Street		NUMBER  10. SPONSOR/MONITOR'S ACRONYM(S)
California Institute of Technology Pasadena CA 91125  9. SPONSORING / MONITORING AGENCY AFOSR/NA 801 North Randolph Street Room 732		10. SPONSOR/MONITOR'S ACRONYM(S)  11. SPONSOR/MONITOR'S REPORT
California Institute of Technology Pasadena CA 91125  9. SPONSORING / MONITORING AGENCY AFOSR/NA 801 North Randolph Street		NUMBER  10. SPONSOR/MONITOR'S ACRONYM(S)
California Institute of Technology Pasadena CA 91125  9. SPONSORING / MONITORING AGENCY AFOSR/NA 801 North Randolph Street Room 732		10. SPONSOR/MONITOR'S ACRONYM(S)  11. SPONSOR/MONITOR'S REPORT
California Institute of Technology Pasadena CA 91125  9. SPONSORING / MONITORING AGENCY AFOSR/NA 801 North Randolph Street Room 732	NAME(S) AND ADDRESS(ES)	10. SPONSOR/MONITOR'S ACRONYM(S)  11. SPONSOR/MONITOR'S REPORT

Approved for public release; distribution is unlimited

#### 13. SUPPLEMENTARY NOTES

#### 14. ABSTRACT

A semi-custom Nd:YAG pump and dye laser system was acquired for implementation of planar laser-induced fluorescence (PLIF) as one of the diagnostic techniques to be used in variable-pressure, hydrocarbon flame experiments. The research will include studies of hydrocarbons, both pure and mixtures, in a pressure range from 0.5 atm to 12 atm, through, but not limited to, two-dimensional imaging of CH and OH radical concentration profiles. The experimental data combined with numerical simulations will allow for the determination of flame properties such as ignition and extinction strain rates, laminar flame speeds, as well as the validation of chemical kinetic schemes in variable-pressure conditions.

#### 15. SUBJECT TERMS

Excimer Laser Laser-Induced Fluorescence Hydrocarbon Combustion

16. SECURITY CLASSIFICATION OF:		17. LIMITATION OF ABSTRACT	18. NUMBER OF PAGES	19a. NAME OF RESPONSIBLE PERSON Julian M. Tishkoff	
a.REPORT Unclassified	b.ABSTRACT Unclassified	c. THIS PAGE Unclassified	υL	6	19b. TELEPHONE NUMBER (include area code) (703) 696-8478

#### Abstract

A semi-custom Nd:YAG pump and dye laser system was acquired for implementation of Planar Laser Induced Fluorescence (PLIF) as one of the diagnostic techniques to be used in variable-pressure, hydrocarbon flame experiments. The research will include studies of hydrocarbons, both pure and mixtures, in a pressure range from 0.5 atm to 12 atm, through, but not limited to, two-dimensional imaging of CH and OH radical concentration profiles. The experimental data combined with numerical simulations will allow for the determination of flame properties such as ignition and extinction strain rates, laminar flame speeds, as well as the validation of chemical kinetic schemes in variable-pressure conditions.

## 1. Equipment description

The laser system is a semi-custom configuration. It was supplied by Spectra-Physics and is comprised of a pump Nd:YAG laser and a multi-stage dye laser.

#### 1a. Pump laser

The Nd:YAG laser (Spectra-Physics, model PRO 290-50) includes the laser head, power supply, and remote-control module. The PRO 290-50 is configured for operation at 50Hz with an optics upgrade (NSI-5) for high-energy operation at 10Hz. It was supplied with the necessary dichroic separation optics to extract 532nm (beam-splitter assembly HIS-532) and 355 nm (beam-splitter assembly OP-355) output. Furthermore, it came with an internal water-cooled beam-dump (BD-6) and was upgrated with a high-efficiency harmonic generator (HG-4B1).

### 1b. Dye laser

The dye laser (Sirah for Spectra-Physics, model PrecisionScan-G) includes the laser head, two sets of associated dye circulators, and computer software for dye-laser control. The PrecisionScan-G was configured with single-grating broadband-linewidth resonator (LG), a preamplifier and one main amplifier; it was upgraded with a double-grating, narrow-linewidth sub-assembly (DG) and a second amplifier for high-energy operation, for a total of three dye cells.

The dye laser includes a wavelength-separation unit (WS-M), a frequency-conversion unit (basic FCU), and two harmonic-generation crystals for broad wavelength operation from 215 to 400nm (basic SHG 260 and upgrade SHG 220). Furthermore, the PresisionScan-G was upgraded with a closed-loop wavelength-stabilization system.

The system was ordered with one year of warranty. Components and upgrades were delivered, installed and tested (February 2000), except for the closed-loop stabilization system (delivery pending).

The original proposal specified an excimer-laser pump, followed by a dye laser. In reviewing possible laser systems for the intended experiments, it was decided that a higher-energy Nd:YAG pump laser, operating at either 10Hz, at high energy, or 50Hz, at lower energy, was better suited for the experiments. These two frequencies are well matched to the Cassini and KFS digital-imaging systems developed in-house under previous and current AFOSR support. The difference in the acquisition cost was cofunded from AFOSR Grant F49620-98-1-0052.

<sup>&</sup>lt;sup>1</sup> DURIP Grant No. F49620-95-1-0199 and AFOSR Grant Nos. F49620-94-1-0353 and F49620-98-1-0052.

## 2. Research use of acquired equipment

The system described above is presently installed and tested. It is being integrated with an experimental configuration for the study of hydrodynamic, pressure, and chemical-kinetic effects on hydrocarbon flames whose goals include the validation of chemical-kinetic models of, initially, low-C hydrocarbon combustion.

Pending support for a major effort in this area (AFOSR proposal submitted), a parametric experimental study combined with extensive numerical simulations will determine the salient features of hydrocarbons throughout their ignition/extinction Z-curve. Flame properties such as ignition and extinction strain-rates as well as flame speeds will be determined as a function of:

- pressure (0.5 to 12 atm),
- fuel type,
- equivalence ratio, and
- ignition temperature.

Associated direct numerical simulations will be conducted along the stagnation streamline (1-D) for the full Navier-Stokes and energy equations, and will include detailed chemistry and detailed transport descriptions for multi-component chemically reacting systems. The experimental and numerical results will generate a comprehensive data library of flame properties of hydrocarbons at normal and elevated pressures that will allow for the validation of chemical kinetic schemes.

The recently acquired laser system is designed for operation at 258, 283, 308 and 389nm. These wavelengths have been documented [e.g., Refs. 1-3] for UV excitation of hydrocarbon-flame radicals such as CH, OH and others in this wavelength range.

Two-dimensional imaging of CH and OH radical concentration profiles will be achieved with a detector system that will use 1024x1024 low-noise, high framing-rate CCD camera systems (Cassini or KFS, depending on laser pulse rate) developed in-house and undergoing continuous improvements.

In work in progress, preliminary experiments were performed to investigate several aspects of the future experimental effort. More specifically, computer-controlled software drivers were developed and tested extensively for an automated experimental procedure. This procedure is designed such that for each fuel type and fuel concentration, all information for ignition, extinction and flame propagation will be measured in a few seconds (20-40s). This approach will allow the time required to create a complete data base for each fuel to be reduced two to three orders of magnitude, when compared the traditional techniques employed, while providing excellent spatial and temporal resolution.

Sensitivity analysis was performed, through direct numerical simulations, to investigate the dependence of the spatial response of the radical concentration profiles to experimental parameters such as pressure, jet velocity, stagnation temperature, reduced

kinetic rates, etc. It was found that there is a significant dependence to these characteristics.

Subsequently, theoretical and numerical calculations were performed for the optimization of the burner design and the accurate determination of hydrodynamic parameters such as velocity and velocity gradients, boundary layer and temperature corrections, etc. The experimental apparatus was optimized such that the overall accuracy of the acquired data will be in the range of 1-3%.

The optical and detection systems used for the radical imaging were designed and optimized to ensure the requisite resolution for the determination of the radical species concentration profiles.

Currently, exploratory experiments are in progress for the UV excitation of CH radicals in stagnation-type of flow of laminar hydrocarbon flames.

### 3. References

- 1. Donbar, J. K., Driscoll J. F. and Carter, C. D., "Structure of the reaction zones within Turbulent Non-premixed Jet flames From CH-OH PLIF images," submitted (1999).
- 2. Donbar, J. K., Driscoll J. F. and Carter, C. D., "Strain rates Measured along the wrinkled flame contour within turbulent non-premixed jet flames," submitted (1999).
- 3. Seitzman, J. M., Miller, M. F., Island, T. C. and Hanson, R. K., "Double pulsed imaging using simultaneous OH/acetone PLIF for studying the evolution of high-speed, reacting mixing layers," *Twenty-Fourth Symposium (International) in Combustion/The Combustion Institute*, 1743-1750 (1994).